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U.S. ARMY MATERIALS RESEARCH AGENCY  
WATERTOWN, MASSACHUSETTS 02152

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# WATERTOWN JOURNAL LADORA

## EXPERIMENTAL

NO. 1

STRESS IN GUNS UNDER COMPRESSION  
MATCHING OF THE EXTERNAL PRESSURE  
ELASTIC GUNS UNDER A DISCONTINUOUS  
PRESSURE BY A SIMILAR THEOREM

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Original Drawings of Figures 1 and 2 filed Applied Mechanics Branch.

Watertown Arsenal Laboratory  
Report Number 730/137-9  
Problem Number L-7.2  
(Partial Report)

21 October 1946

SUBJECT

Stresses in Gun Tubes  
Stresses in Guns under Combined Band and Gas Pressures, Part 10,  
Matching of the External Tangential Strain Data in Smooth-bored  
Elastic Guns under a Discontinuous Band of Internal Radial  
Pressure by a Simple Mathematical Expression

OBJECT

To present a simple mathematical expression from which the  
external tangential strains in smooth-bored elastic guns under a  
discontinuous band of internal radial pressure may be calculated.


SUMMARY


The external tangential strains in smooth-bored gun tubes loaded  
by a semi-infinite band of internal radial pressure may be represented  
by an expression of the form

$$e^{-\alpha z} \{ C_1 \cos \beta z + C_2 \sin \beta z \}$$

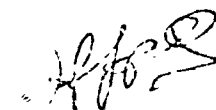
where  $\alpha$ ,  $\beta$ ,  $C_1$  and  $C_2$  are constants varying with wall ratio of the  
tube. Curves of these constants plotted versus wall ratio are  
contained in Figures 1 and 2. For wall ratios of 1.00 to 2.50, values  
of these constants may be found from Figures 1 and 2, and the external  
tangential strains may be computed. The derivatives of the external  
tangential strains may of course be found similarly by using the given  
values of  $\alpha$ ,  $\beta$ ,  $C_1$  and  $C_2$  and using the differentiated form of the  
above expression.

APPROVED:

  
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## DISCUSSION

The elastic strains at the outer surface of smooth-bored gun tubes under internal radial pressure loading have been presented in report<sup>1</sup> WAL 730/137-1. A semi-infinite band of pressure of unit magnitude extending over the left side of the infinite tube was taken for the loading distribution. For convenience, the strains were divided by the strain corresponding to a uniform radial pressure as given by the usual Lamé formula for an open-ended tube. The external tangential strains were thus presented in the form

$$\frac{w^2 - 1}{2} E e_t$$

where,

$w$  = wall ratio of tube =  $\frac{\text{outside diameter}}{\text{inside diameter}}$

$E$  = Young's Modulus

$e_t$  = external tangential strain due to the  
semi-infinite band of pressure.

This expression, of course, varies along the axial distance of the tube.  $e_t$  is, therefore, a function of  $z$  where  $z$  is measured in the positive direction to the right of the point of discontinuity of loading.  $z$  is considered negative at all points to the left of the discontinuity of loading.  $z$  is axial distance measured in units of the bore radius.

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1 Report Number WAL 730/137-1, "Stresses in Gun Tubes, Stresses in Guns under Combined Band and Gas Pressures, Part 2, Elastic Strains at the Outer Surface for Internal Radial Pressure, Basic Data."

Dr. R. Beeuwkes, Jr. has presented in a recent report<sup>2</sup> an interesting discussion as to the matching of the function  $e_t$ . He points out that an expression of the form

$$e^{-\alpha s} \{C_1 \cos \beta s + C_2 \sin \beta s\}$$

with the proper choice of the constants  $\alpha$ ,  $\beta$ ,  $C_1$  and  $C_2$  will fit the external tangential strain data.

The external tangential strains may be, therefore, calculated from

$$\begin{aligned} \frac{w^2-1}{2} Ee_t &= e^{-\alpha s} \{C_1 \cos \beta s + C_2 \sin \beta s\} \text{ for } s \geq 0 \\ &= 1 - e^{-\alpha s} \{C_1 \cos \beta s + C_2 \sin \beta s\} \text{ for } s \leq 0 \end{aligned}$$

where values of  $\alpha$ ,  $\beta$ ,  $C_1$  and  $C_2$  are found for any wall ratio from  $w = 1.00$  to  $w = 2.50$  in Figures 1 and 2.

In order to correlate  $\alpha$  and  $\beta$  with thin wall theory,  $\alpha$  and  $\beta$  have been divided by  $\lambda$  where

$$\lambda = \frac{[12(1-\mu^2)]^{\frac{1}{4}}}{[w^2-1]^{\frac{1}{2}}} = \frac{1.8222}{[w^2-1]^{\frac{1}{2}}} \text{ where } \mu = 0.285.$$

From the data in Figures 1 and 2, the values of  $\frac{w^2-1}{2} Ee_t$  will be represented at least up to an accuracy of .005, except for very large values of  $s$ , at large values of  $w$ , where the original data in some places is in error by as much as .020.

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2 Report Number WAL 730/419, "Stresses in Thick Walled Cylinders."

The derivatives of the external tangential strains which appeared in WAL 730/137-7 can, of course, be calculated by using the  $\alpha$ ,  $\beta$ ,  $C_1$  and  $C_2$  values in Figures 1 and 2 and substituting them in the derivative of

$$e^{-\alpha} \{C_1 \cos \beta + C_2 \sin \beta\} .$$

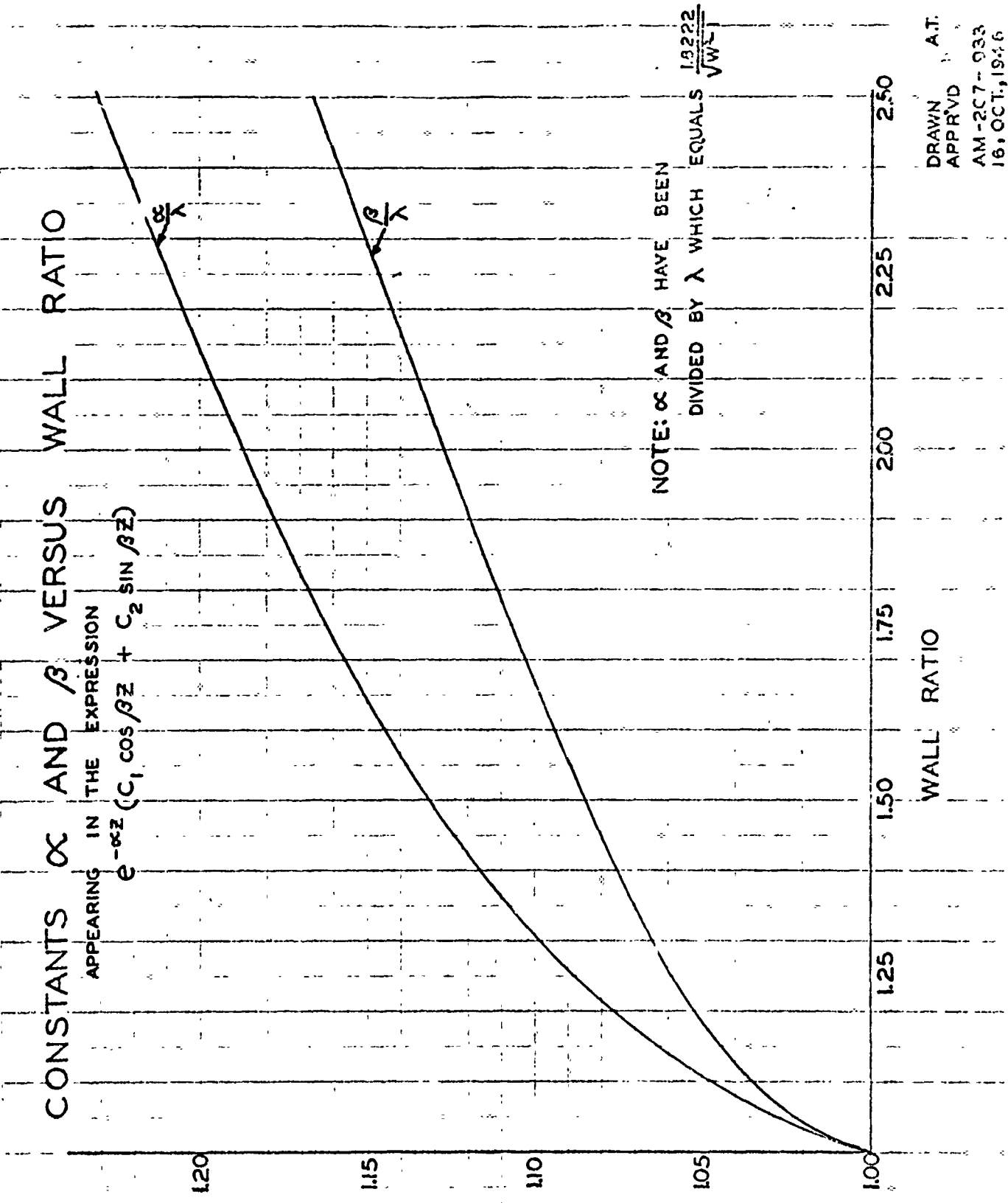
#### ACKNOWLEDGMENT

In addition to the direction and advice of R. Beeuwkes, Jr., much of the actual calculation was performed by A. Tashjian of the Applied Mechanics Branch, Watertown Arsenal Laboratory.



# CONSTANTS $\alpha$ AND $\beta$ VERSUS WALL RATIO

APPEARING IN THE EXPRESSION  
 $e^{-\alpha z} (C_1 \cos \beta z + C_2 \sin \beta z)$



NOTE:  $\alpha$  AND  $\beta$  HAVE BEEN  
 DIVIDED BY  $\lambda$  WHICH EQUALS  $\frac{1.9222}{\sqrt{W/L}}$

DRAWN A.T.  
 APPR'D  
 AM-207-933  
 16, OCT., 1946

# CONSTANTS $C_1$ AND $C_2$ VERSUS WALL RATIO

APPEARING IN THE EXPRESSION

$$e^{-\alpha z} (C_1 \cos \beta z + C_2 \sin \beta z)$$

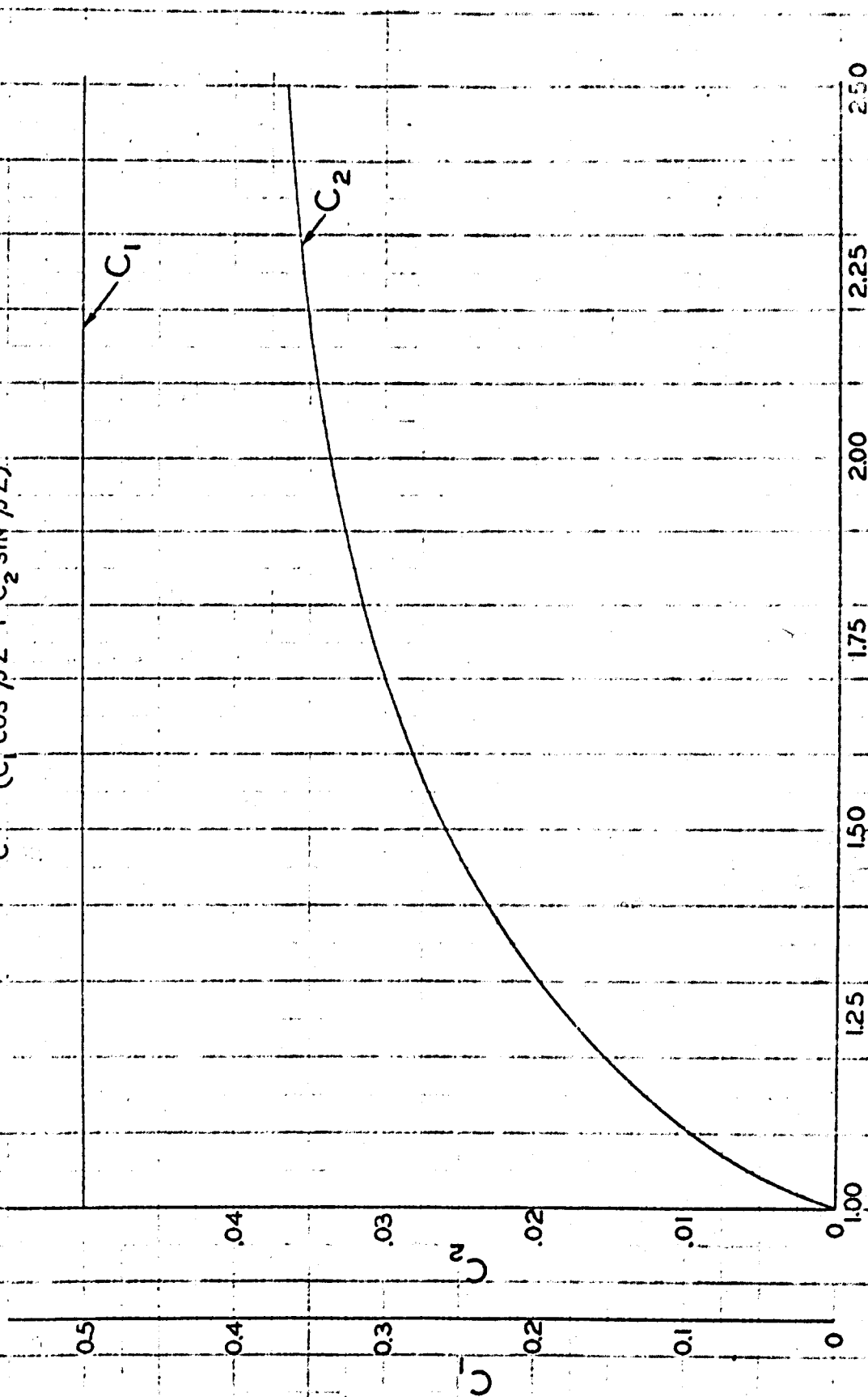


FIG. 1  
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